

Article

Compactibility and Fibre Volume Fraction Limits of Unidirectional Discontinuous Carbon Fibre Thermoset Prepreg Laminates

Miriam Preinfalck^{1,*}, Julian Kupski², Mohammad Hajikazemi² , Christian Brauner² , Stephan Baz¹ and Götz T. Gresser¹

¹ Deutsche Institute für Textil- und Faserforschung, Korschtalstraße 26, 73770 Denkendorf, Germany; stephan.baz@ditf.de (S.B.); goetz.gresser@ditf.de (G.T.G.)

² Institute of Polymer Engineering, FHNW University of Applied Sciences and Arts Northwestern Switzerland, Klosterzelgstrasse 2, 5210 Windisch, Switzerland; julian.kupski@fhnw.ch (J.K.); mohammad.hajikazemi@fhnw.ch (M.H.); christian.brauner@fhnw.ch (C.B.)

* Correspondence: miriam.preinfalck@ditf.de

Abstract

The aim of this study was to explore the compactibility of unidirectional staple carbon fibre laminates in comparison with their uni- and biaxial continuous fibre counterparts. Resin-preimpregnated plies were inserted into a heated compression mould at an elevated mould temperature of 110 °C. By applying stepwise loading, the correlation between consolidation pressure and fibre volume content was derived and related to fibre orientation distribution. The fibre orientation distribution is obtained from photographic image analyses of 2D ply sections of the same samples using the structure tensor approach. For commonly used autoclave prepreg pressure of 6.8 bar results indicate that lower-oriented staple carbon fibre unidirectional laminates with a fibre orientation distribution factor $\eta_0 = 0.74$ can potentially reach a maximum of 39% fibre volume fraction, while higher-oriented laminates with $\eta_0 = 0.78$ end up at 43%. An exponential extrapolation suggests that a consolidation pressure of ≥ 90 bar is required to achieve 60% fibre volume content with highly oriented unidirectional staple carbon fibre laminates.

Keywords: staple carbon fibre; fibre orientation distribution; ply compactibility; unidirectional laminates; fibre volume content; thermoset prepreg; recycled carbon fibre

1. Introduction

Carbon fibre reinforced polymers (CFRPs) enable outstanding specific stiffness and strength and are therefore widely used in lightweight structural applications. However, increasing production volumes and limited recyclability of thermoset matrices have intensified the demand for resource-efficient material solutions. The reuse of carbon fibres, particularly in the form of short, recycled staple carbon fibres (CF), has thus emerged as a promising strategy to reduce environmental impact while retaining high mechanical performance [1].

Staple fibre composites consist of discontinuous fibres typically ranging from millimetres to centimetres in length. Compared with continuous fibre systems, they offer broad material utilisation, lower cost, and greater flexibility in processing [2,3]. However, fibre discontinuity introduces microstructural heterogeneity, fibre-end gaps, and imperfect alignment, all of which influence mechanical performance and consolidation behaviour.

Carding is commonly employed to align staple fibres prior to tape or yarn formation. During this process, fibres are carded (oriented by means of metal pins, similar to



Academic Editors: Juan Carlos Merino, Mercedes Santiago-Calvo, María Asensio-Valentin and Karina Nuñez

Received: 25 April 2026

Revised: 29 May 2026

Accepted: 4 June 2026

Published: 12 June 2026

Copyright: © 2026 by the authors. Licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the [Creative Commons Attribution \(CC BY\) license](https://creativecommons.org/licenses/by/4.0/).

brushing), aligned, and formed into a uniform web, which is then condensed to slivers and subsequently into yarn or tapes for composite manufacturing [4–7]. The resulting fibre architecture governs both the load transfer efficiency and the packing behaviour during consolidation. Consequently, controlling fibre orientation distribution (FOD) is essential for achieving structural-grade performance [8,9].

Besides the material properties of the constituents and the fibre-matrix interface properties [10,11], the mechanical response of fibre-reinforced composites is determined by microstructural descriptors such as fibre orientation [12–15], fibre length [16,17], interfibre spacing [18], fibre diameter [19], fibre connectivity and curvature [20], fibre volume content (FVC) [21], and porosity [22,23]. These features are quantified using destructive methods (optical microscopy [24,25], mechanical testing [7,26,27]) and non-destructive techniques including X-ray diffraction [28], X-ray computed tomography [18,20], ultrasound-based methods [29,30], and spectroscopic approaches [31]. Image analysis techniques such as Fourier transforms and structure tensor methods enable statistical evaluation of FODs [15,32]. Moreover, micromechanical models frequently employ modified rule-of-mixtures formulations to relate microstructure to elastic properties. However, such models typically assume a known and uniform FVC. In staple fibre laminates, the attainable FVC is itself a function of fibre architecture and consolidation pressure, introducing a coupling between microstructure and process mechanics that is often neglected.

In unidirectional (UD) continuous fibre laminates, fibres are ideally aligned and assumed infinitely long. In contrast, staple fibre laminates exhibit both finite fibre length and a distribution of fibre orientation angles [20]. These effects can be incorporated into the longitudinal modulus prediction using efficiency factors. These efficiency factors were implemented for oriented staple fibre architecture laminae [24,33,34]:

$$E_{11} = \eta_0 \eta_1 V_f E_f + V_m E_m \quad (1)$$

where E_{11} is Young's modulus of the composite; η_0 is a correction factor for the fibre orientation; η_1 is a correction factor accounting for the overlap length between adjacent fibres, based on the Cox model [35]; V_f and E_f are the fibre properties; and V_m and E_m are the matrix properties. η_0 is defined as

$$\eta_0 = \sum \beta_i \cos^4 \theta_i \quad (2)$$

with β_i denoting the frequency of fibres oriented at angle θ_i . For perfectly aligned fibres, $\eta_0 = 1$. For fibre lengths significantly exceeding the critical transfer length ($\gg 1$ mm in typical carbon/epoxy systems), $\eta_1 \approx 1$, indicating that stiffness reduction is dominated by orientation rather than shear-lag effects. From a tensorial perspective, η_0 corresponds to a scalar representation of the second-order orientation tensor projected onto the loading direction. Beyond stiffness reduction, fibre misalignment also affects packing density. Increasing angular deviation increases excluded volume and limits fibre nesting during compaction. Thus, orientation distribution simultaneously governs mechanical efficiency and achievable fibre volume fraction.

The compaction of fibre beds can be described as a pressure-dependent reduction in void space within a granular-like network of interacting filaments or fibres. For discontinuous fibres, three mechanisms dominate:

- (i) Fibre reorientation and nesting at low pressures;
- (ii) Bending and local rearrangement at intermediate pressures;
- (iii) Fibre deformation (up to fibre breakage) and matrix-dominated compression at high pressures.

Staple fibre preforms are generally more voluminous than continuous fibre laminates due to fibre-end gaps, waviness (undulation and crimp), degree of fibre orientation and imperfect stacking. Even in theoretically aligned systems, fibre ends introduce unavoidable geometric inefficiencies. The shorter the fibres, the higher the frequency of end gaps and the lower the maximum attainable packing density [36]. From practical manufacturing experience, achieving FVC \approx 60% under autoclave pressures of 6–7 bar is challenging for staple fibre laminates. Misaligned fibres increase steric hindrance and cause premature densification plateaus, requiring substantially higher consolidation pressures compared with continuous fibre systems.

Edwards and Evans [36] first quantified the pressure–volume relationship of aligned short fibre mats. Centrifugally aligned 3 mm CF mats achieved FVC \approx 50% at 7 bar and \sim 60% at 69 bar. While continuous fibre mats approached \sim 70% FVC, discontinuous systems were theoretically limited to \sim 47% at low pressure due to fibre-end effects, highlighting intrinsic packing constraints. Eom et al. [37] linked fibre bed compressibility to cure-induced stress, identifying a consolidation window of 1–5 bar for achieving 50–62% FVC without void formation. Likewise, Kirupanantham [34] reported non-linear densification of discontinuous carbon fibre laminates, with compaction plateauing above 20 bar and final FVC values of \sim 55–58%, indicating structural packing limits rather than insufficient pressure. Subsequent studies focused on improving the understanding and characterisation of fibre bed compaction. Experimental and modelling approaches demonstrated reliable prediction of laminate densification [38], while investigations of bundle mechanics and packing behaviour highlighted the influence of fibre architecture, needling, and sizing on compaction response and moulding performance [39,40]. At the same time, work on testing methodology addressed inter-laboratory variability, machine compliance, and in situ characterisation to improve the accuracy and process relevance of compaction measurements [41–43].

Although the compaction behaviour of textile reinforcements and continuous fibre systems is well established, quantitative relationships between FOD and achievable FVC in staple CF prepregs remain unresolved. Existing studies typically characterise material properties at comparatively low FVC and extrapolate to higher compaction levels, despite limited evidence that such FVCs are attainable under realistic consolidation pressures. A predictive framework linking η_0 , consolidation pressure, and resulting FVC under industrially relevant autoclave conditions is currently absent. Resolving this gap is critical for defining manufacturable design windows and for assessing whether improvements in fibre alignment can offset pressure limitations in industrial processing. Accordingly, this study quantifies for the first time the compactibility of UD staple CF thermoset prepreg laminates relative to continuous fibre systems and establishes a mechanistic relationship between FOD and achievable FVC as a function of consolidation pressure.

2. Materials and Methods

2.1. Materials

The methodology was demonstrated using two types of in-house manufactured UD staple CF tapes with varying alignment processing parameters, designated as low-stretch (LS) and high-stretch (HS). The average length of a single fibre in these tapes was approximately 50 mm, and the average fibre diameter was 7 μ m. The fibre used in all samples was of the standard modulus/high-strength type, T700SC-12K-50C, following the naming convention of Toray Industries, Inc. (Tokyo, Japan). Alongside the staple CF tapes, two reference samples were produced using the same fibres in a dry UD carbon layup with 80 g/m². Table 1 depicts an overview of the layup and the areal weight of all configurations, taking into account the sizing amount on the chosen fibre and a small deviation in added

binder between staple and continuous fibre preforms. Figure 1 shows optical images of both staple fibre tape configurations, HS and LS, utilised in this study.

Table 1. Laminate properties of tested coupons with Torayca T700SC 12K-50C.

	Layup	Binder + Sizing Portion [m%]	Dry Areal Weight [g/m ²]
UD staple CF—LS	[0] ₁₂	11.05	80
UD staple CF—HS	[0] ₁₂	11.05	80
UD continuous CF	[0] ₁₂	5.34	83.5
Bi-axial continuous CF	[0, 90] _{3s}	5.34	83.5

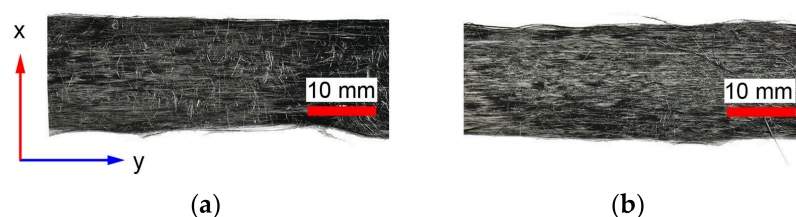


Figure 1. Top-view images of the tapes utilised in this study depicting: (a) low-stretch (LS) and (b) high-stretch (HS) tape, both with a nominal tape width of 16 mm [15].

A total of 12 layers of staple CF tapes, as well as reference continuous CF tapes of 83.5 g/m² areal weight, were laminated to form flat test specimens. Resin films with an areal weight of 100 g/m² were produced using the HLCL-1000 hot melt laboratory coater from ChemInstruments (Fairfield, OH, USA). These resin films were then used to manufacture prepreps by film-stacking them with the tapes in the desired layup configuration. To prevent excessively rapid curing of the resin, only the EPIKOTE™ Resin 05545 (Westlake Corporation, Houston, TX, USA) component was employed as a film to the dry tapes. Consequently, potential sources of variability associated with curing reactions, such as mixing ratio deviations, pot-life effects, or exothermic heat generation, were excluded. A preliminary rule-of-mixtures calculation indicates that the areal weight of the supplied resin film would exceed the amount typically required to achieve FVC in the range of 50–60%. This was intentionally selected to ensure an adequate resin supply during testing and to minimise the risk of void formation. The tapes were carefully aligned in parallel and accurately cut to match the dimensions of the mould cavity. Prior to each testing cycle, the mould cavity and stamp were coated with Zyvax® WaterShield™ EU (Chem-Trend, Howell, MI, USA) as a release agent.

2.2. Mesoscopic FOD Analysis Method

In this study, the FOD analysis technique based on photographic imaging of each ply of the laminate was chosen. This method was previously introduced by Zweifel [15], and the present work aims to link his past FOD visual inspection results with novel compaction behaviour curves for the same samples. Image analysis based on 2D ply sections was performed using the structure tensor approach to characterise FODs. The structure tensor is a matrix that describes the local second-order structure of an image and is often used in computer vision and image processing for edge detection, texture analysis and directionality. Furthermore, the degree of homogenisation or representative volume element (RVE) was addressed in the frame of different region of interest (ROI) volumes.

FODs were determined using a two-dimensional structure tensor approach implemented in Python (v3.8.12). During plate manufacturing, each ply was photographed with a calibrated DSLR setup (Nikon D810, Nikon Corporation, Tokyo, Japan) equipped

with a 50 mm lens and a circular polarisation filter to minimise reflection artefacts. Images ($5'759 \times 2'879$ pixels, 300 dpi) were corrected for lens distortion using OpenCV [13]. Automated image processing subdivided the images into block segments for spatial homogenisation. Fibre orientations were extracted via a convolution-based structure tensor method and orientation histograms were converted into probability density functions. The alignment coefficient η_0 was calculated for each segment. Virtual stacking of the plies yielded laminate-level median and standard deviation values of η_0 for each configuration. Using this method, it was revealed that most fibres that are not correctly aligned appear at the ply surface or boundaries. Thus, 2D mesoscopic analysis can be utilised to identify critical fuzzy regions in the tape.

2.3. Specimen Manufacturing and Testing

A square consolidation mould with a $32 \times 32 \text{ mm}^2$ cavity was developed for integration into a universal testing machine (K100, Zwick Roell GmbH & Co. KG, Ulm, Germany), enabling controlled stepwise hydrostatic compaction and precise thickness measurement at elevated temperature [44]. Tests were conducted at $110 \text{ }^\circ\text{C}$. This processing temperature was defined according to Walker [45], who identified $110 \text{ }^\circ\text{C}$ as optimal for binder dissolution from the staple CF tapes into the epoxy matrix within several minutes. At lower temperatures dissolution remains incomplete, whereas at higher temperatures resin curing may outpace binder diffusion, potentially influencing consolidation behaviour.

The lower mould assembly consisted of a heated frame mounted on a water-cooled base plate and supported by a rigid steel backing plate to minimise compliance. The upper assembly comprised a water-cooled stamp with integrated heating, connected to the load cell via a cooled adapter (Figure 2).

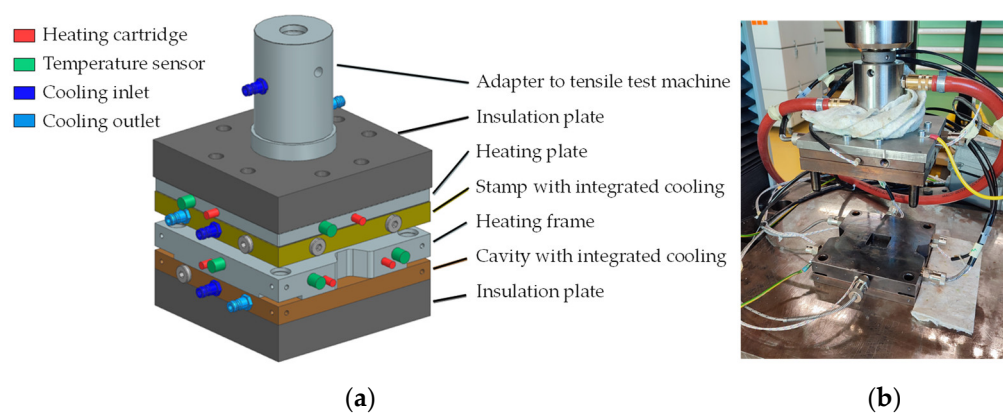


Figure 2. Consolidation experiment setup [44]: (a) schematic illustration; (b) photograph of the setup installed in the testing machine.

Machine compliance can significantly affect compaction measurements if not properly corrected, as demonstrated by Sousa et al. [42]. In the present setup, ceramic insulation plates were identified as the primary source of compliance and were therefore removed. Thermal control was ensured through enhanced active cooling of the upper mould and the addition of a rigid base plate to moderate heat transfer (Figure 2). Residual system deformation was quantified by performing empty mould runs, which were subtracted from the measured specimen displacement. All tests were executed after passing a 30 min heat-up of the entire test setup to ensure thermal equilibrium and avoid displacement artefacts caused by transient thermal expansion of the tooling.

In accordance with Kirupantham [34], compaction was treated as a time-dependent process governed by resin flow and viscoelastic relaxation. Under constant load, laminate thickness continued to decrease due to resin redistribution within the fibre network, partic-

ularly at low consolidation pressures. The load was applied incrementally at a crosshead speed of 0.2 mm/min in 50 N steps, from an initial load of 50 N up to 2000 N (40 load steps). The selected number of load increments represents a compromise between achieving sufficient data resolution for deriving pressure–FVC correlation curves and limiting overall test duration. Prolonged testing may lead to gradual thermal drift of the moulding setup despite the integrated cooling system, potentially affecting calibration and displacement accuracy. At each level, displacement was held constant for at least 300 s to allow force relaxation and to ensure quasi-equilibrium thickness before proceeding to the next increment. The resulting load over time/displacement data, as displayed in Figure 3, were used to derive correlation curves linking consolidation pressure to achievable FVC. The FVC was thereby calculated using the following expressions:

$$\text{FVC} = \frac{V_f}{V} \quad V_f = \frac{m_f}{\rho_f} \quad V = A \times \Delta h \quad (3)$$

with Δh corresponding to the measured displacement of the testing machine, A being the mould surface area, and $\rho_f = 1.79 \text{ g/cm}^3$, representing the CF density.

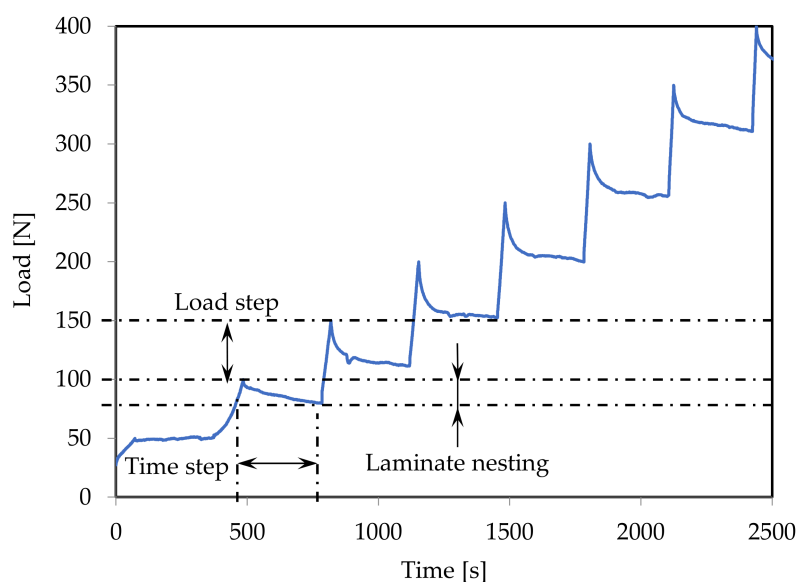


Figure 3. Extract from the raw data curve: load over time with load reduction after 300 s at constant displacement.

The gap between the upper and lower mould enabled excess low-viscosity epoxy to flow out of the cavity during processing. Post-test observations show significant resin accumulation outside the cavity, consistent with the initially supplied excess resin, indicating that sufficient drainage occurs and that the measured thickness is primarily governed by the fibre preform.

3. Results

3.1. FOD Analysis Results

Figure 4 presents the results of the FOD analysis. Zweifel [15] measured $\eta_0 = 0.741$ for the LS and 0.781 for the HS tape configuration. In comparison, a UD continuous CF non-crimped fabric was measured with $\eta_0 = 0.934$. These measurements were taken on dry preforms prior to consolidation.

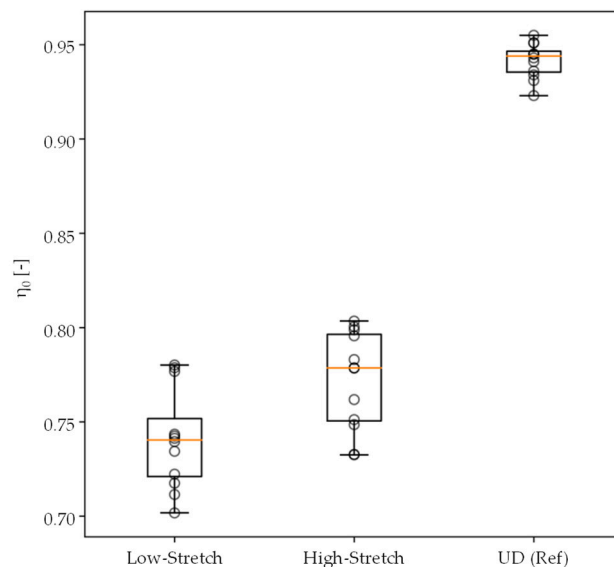


Figure 4. Derived η_0 values from virtually stacked optical images acquired ply-by-ply with a Digital Single-Lens Reflex (DLSR) camera.

3.2. Derivation of Pressure over FVC Curves

The calculated compaction curves relating mould pressure to FVC are presented in Figure 5. Continuous CF laminates exhibit significantly higher compactibility under moderate pressure compared with staple CF systems, even for biaxial layups. In continuous UD and biaxial configurations, efficient intra-ply filament nesting enables rapid densification at low pressures. In contrast, UD staple CF laminates display reduced packing efficiency due to fibre discontinuity and misalignment. Notably, substantial differences were observed between staple CF specimens manufactured from the same base material. The HS configuration exhibited increased scatter in the pressure–FVC correlation.

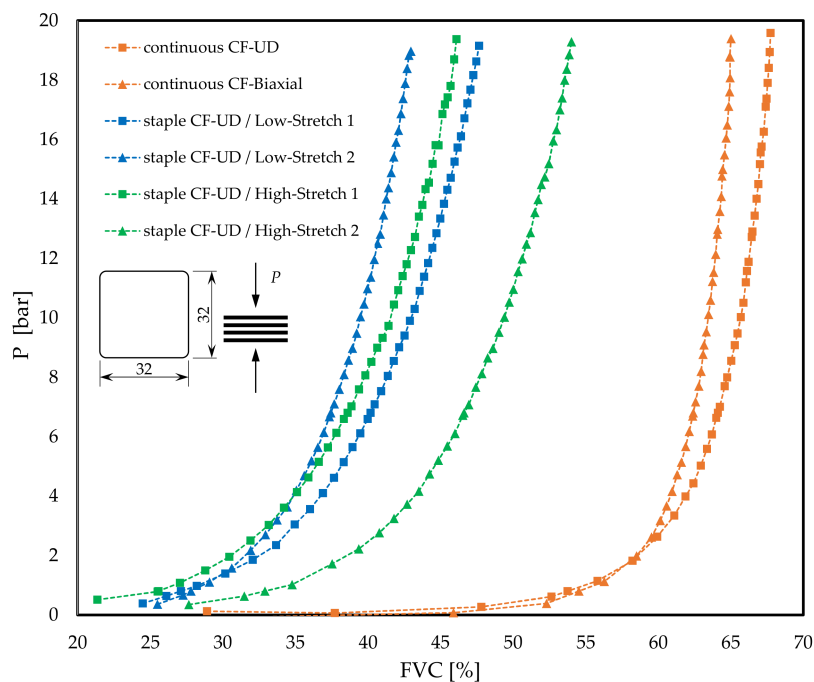


Figure 5. Cavity pressure over FVC for UD staple and continuous CF tapes.

The observed behaviour is consistent with trends reported by Kirupanantham [34] and Grieder [44], showing non-linear densification with a pronounced plateau at higher pressures.

This variability is attributed to regional FOD differences and local thickness variations, as previously indicated by Zweifel [15], see Figure 6. There, the most representative segment was chosen by fitting a Weibull distribution to the apparent fibre orientation correction factor derived from histograms (structure tensor). A Weibull distribution was chosen because the average value does not represent the data well.

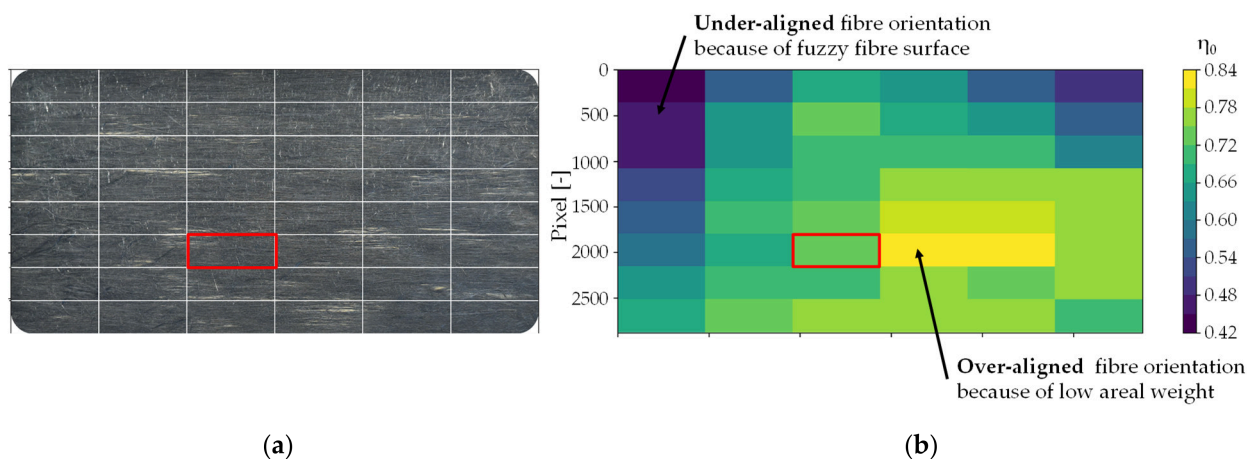


Figure 6. Over- and under-alignment regions on scanned layer surfaces. (a) photograph of a ply using a polarisation filter; (b) false-colour fibre orientation map.

These inhomogeneities originate from the carding and the subsequent alignment process of the staple fibres to tapes as well as from the alignment of the tapes. Variations in the initial fibre length distribution extend beyond the well-characterised profiles produced by modern processing machinery [2,3]. These fibre length variations contribute to non-uniform sliver formation, leading to localized regions enriched in either shorter or longer fibres. During subsequent alignment, due to the big difference between the shortest and longest fibres, the uniformity variations intensify, leading to areas of increased fibre-end density and locally thicker tape sections, alongside regions dominated by longer fibres that appear comparatively thinner. Such spatial heterogeneity directly affects compaction response and contributes to the observed scatter in FVC.

3.3. Data Extension to Distinct Process Parameters

Linear interpolation and exponential extrapolation of the compaction curves were performed to provide an indicative estimate of FVC under typical processing conditions, presented in Figure 7 and Table 2. The extrapolation was done without the introduction of formal uncertainty bounds. For vacuum processing (0.8 bar), predicted FVC values reach 27.4% (LS) and 29.1% (HS). Under autoclave conditions (6.8 bar), FVC increases to 38.8% (LS) and 42.6% (HS).

For the HS configuration, achieving an aerospace-grade FVC of approximately 60% appears to require pressure levels on the order of ~ 90 bar. Hereby, two visually distinct parts of the curves in Figure 7 represent the two different data types: The first segment (lines with markers) corresponds to the experimentally measured compaction data consisting of 40 discrete load steps. The second segment (continuation of lines without markers) represents the exponential fit applied to the experimental data.

However, this estimate should be interpreted cautiously, since it extends beyond the experimentally investigated pressure range and is influenced by experimental scatter and local laminate heterogeneity. For the LS configuration, the required pressure would theoretically exceed this range even more substantially. The extrapolation serves primarily

to illustrate the practical limitations of pressure-driven densification in the investigated staple CF system rather than to provide an exact predictive threshold.

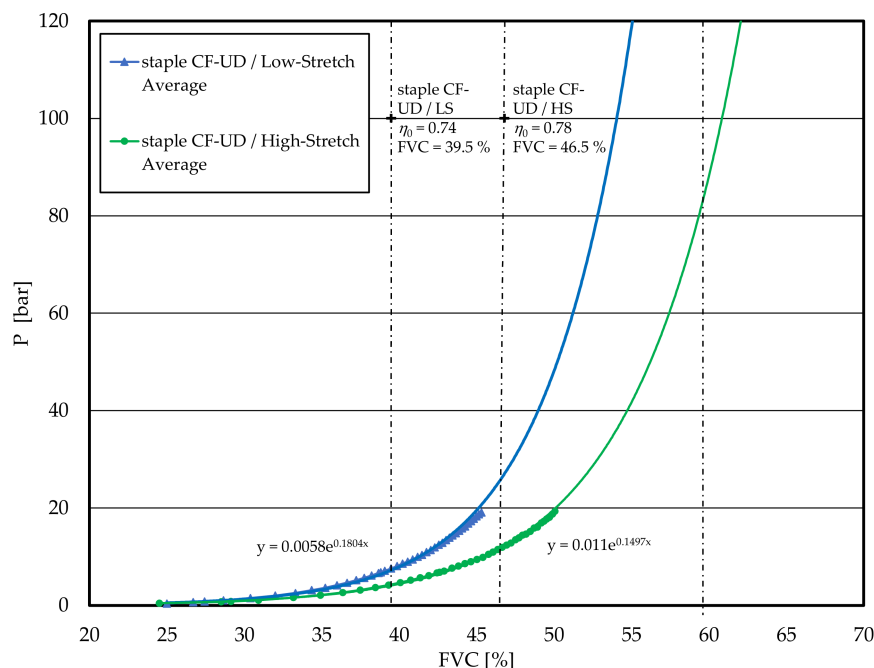


Figure 7. Averaged pressure–FVC curves for LS and HS configurations, exponentially extrapolated.

Table 2. Inter- and extrapolated FVC for typical prepreg consolidation processes.

	P [bar]	FVC [%]
Vacuum processing/LS	0.8	27.4
Autoclave processing/LS	6.8	38.8
Aerospace grade/LS	n/a	60.0
Vacuum processing/HS	0.8	29.1
Autoclave processing/HS	6.8	42.6
Aerospace grade/HS	~90	60.0

Another 60 under reasonable autoclave processing conditions (~6 bar). To obtain this information, the data in Figure 5 were combined with FOD information expressed in terms of η_0 and a surface was fitted to obtain an interpolated function of the pressure required as a function of FVC and η_0 . For this curve to be highly precise, a large number of compaction tests would likely be needed for staple CF composites with different FOD levels, which does not seem feasible. Additionally, the same tape was used to manufacture continuous fibre laminates in both unidirectional and biaxial layups, resulting in similar η_0 values for these two laminate types. Some scatter is observed in tests for both low-stretch and high-stretch samples, even when using a single η_0 to express their in-plane FOD. However, to gain a practical understanding of how FOD affects the achievement of aerospace-grade FVC in realistic autoclave processing, the FVC–pressure results of different staple CF laminates were averaged in Figure 7. A slightly lower value of $\eta_0 = 0.9$ was assigned to the biaxial laminate, and a surface was fitted to express pressure as a function of FVC and η_0 . The resulting surface is shown in Figure 8. Analysis of this figure indicates that, for UD staple CF tapes to reach an FVC of 60% at 5.84 bar, η_0 should be approximately 0.89. This is not an absolutely accurate threshold due to the simplifying assumptions made in constructing

Figure 8, but it provides useful guidance regarding the effort required to use staple fibres in structural applications.

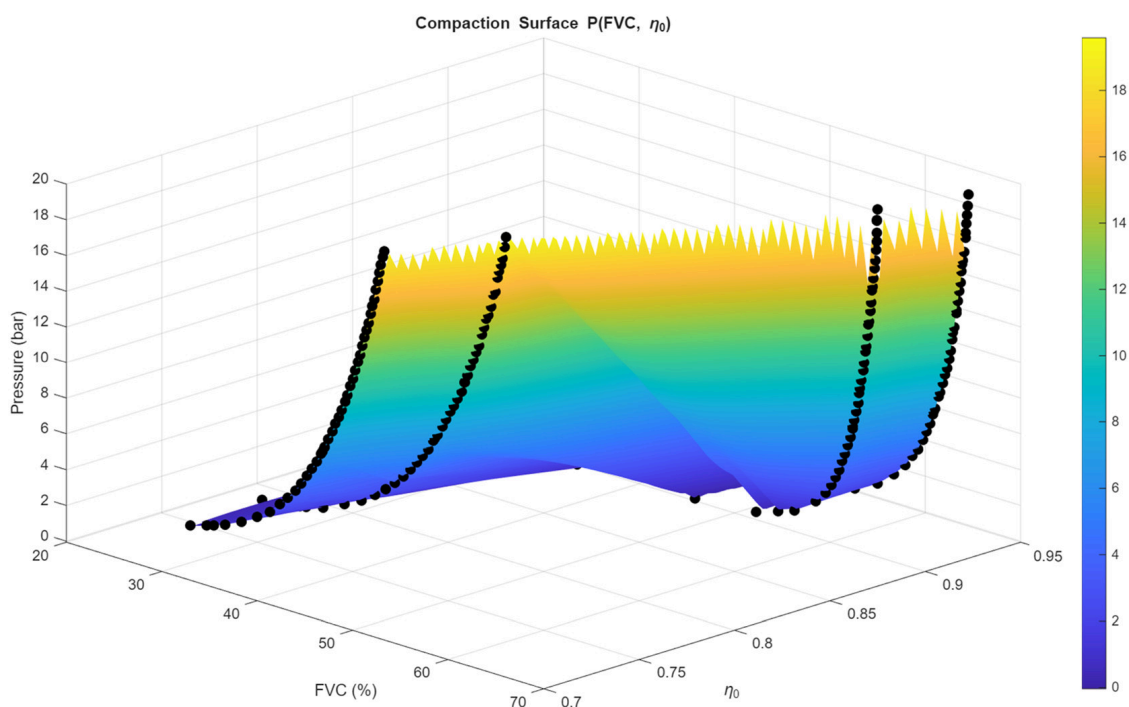


Figure 8. Surface showing the required compaction pressure as a function of FVC and fibre orientation factor (η_0) for staple CF laminates.

Finally, it should be noted that the FOD considered in this study represents the in-plane fibre orientation derived from 2D image analysis. The compactibility may also be influenced by additional microstructural features that are not fully captured by this metric, such as out-of-plane fibre orientations, fibre undulations, non-uniform fibre distribution, and resin-rich regions. These characteristics can further reduce packing efficiency, particularly during the initial stage of compaction often described as the fibre reorientation and nesting phase at low pressures. In this regime, the laminate structure behaves more similarly to a loosely stacked multi-ply system, where limited filament mobility and hindered intra-ply nesting restrict early densification.

3.4. Comparing FVC-FOD Results with Previous Work

Table 3 compares the present results with previously reported data for UD staple CF laminates. Considerable scatter is observed between studies, both in reported alignment coefficients η_0 and achieved FVC values. For example, HiPerDiF systems [22] with fibre lengths of 1–3 mm achieved $\eta_0 \approx 0.90$ – 0.92 and FVC values between 41 and 55%, while the TuFF process [27] reported FVC up to 63% at similar orientation levels. Given 100 bar of consolidation pressure, Walker [45] reported FVC $\approx 53\%$ at $\eta_0 = 0.85$ for 50 mm fibres. With the same processing parameters Zweifel [15] measured lower alignment factors for the same 50 mm staple fibre tapes ($\eta_0 = 0.741$ for LS and 0.781 for HS), corresponding to FVC values of 39.5% and 46.5%, respectively.

The deviation between extrapolated predictions and experimentally obtained FVC values can largely be attributed to spatial FOD inhomogeneities. Over- and under-aligned regions reduce the effective laminate-level η_0 , limiting achievable packing density despite local alignment improvements. Furthermore, direct comparison between studies is complicated by differences in FOD and FVC measurement techniques, as well as variations in fibre length, diameter and surface topology. Zweifel assumed a fictitious fibre length of

10 mm within his structure tensor approach. Experimental measurements after the carding process, however, indicated approximately 50 mm average length. Although fibre length primarily affects the shear-lag efficiency factor η_1 rather than η_0 , it may still influence packing behaviour and thus the attainable FVC through fibre-end density and nesting characteristics. The comparison between studies should therefore be interpreted with caution. Besides differences in FOD and FVC measurement methodologies, variations in fibre length, diameter, surface topology, and processing routes may influence the reported compaction behaviour. In the present work, fibre length was not systematically varied and therefore its individual contribution to compactibility and attainable FVC cannot be isolated from the effects of fibre orientation and regional heterogeneity. Although fibre length may influence packing behaviour through fibre-end density and nesting characteristics, the current dataset does not allow quantitative conclusions regarding this effect. Accordingly, potential fibre-length influences should be regarded as a limitation of the present study and as a subject for future investigation rather than as a demonstrated mechanism.

Table 3. Comparison with FOD/FVC values from the literature.

	Average Fibre Length [mm]	η_0 [-]	FVC [%]
Yu, HiPerDiF [24]	1	0.90	41
Yu, HiPerDiF [24]	3	0.92	55
Yarlagadda, TuFF [27]	5	0.91	63
Walker, FHNW [45]	50	0.85	53
UD staple CF—LS [15]	10/50	0.74	40
UD staple CF—HS [15]	10/50	0.78	47

4. Conclusions

Compaction trials were conducted to determine the maximum practical FVC of UD staple CF thermoset prepreg tapes under varying hydrostatic pressures. Square specimens of $32 \times 32 \text{ mm}^2$ were manufactured from two alignment states (LS and HS) and compared with continuous CF $[0]_{12}$ and biaxial $[0/90]_{3S}$ laminates. Pressure–thickness data were evaluated to derive correlation curves linking consolidation pressure to achievable FVC. These results were subsequently linked to FOD mesoscopic image analysis for the same UD staple CF tapes in dry state.

The following conclusions can be drawn:

- Continuous CF laminates exhibit significantly higher compactibility under moderate pressures compared to UD staple CF tapes. Efficient intra-ply filament nesting enables rapid densification, even for biaxial stackings, whereas fibre discontinuity and misalignment limit packing efficiency in staple systems.
- The compaction response of staple CF laminates is strongly governed by FOD. Both mesoscopic image analysis and mechanical compaction trials indicate that higher nominal alignment (HS) is associated with increased scatter in FVC. This behaviour is attributed to spatial inhomogeneities introduced during the carding and alignment process, leading to locally over- and under-aligned regions.
- Extrapolation of the compaction curves demonstrates that aerospace-grade FVC values (~60%) would require consolidation pressures approaching 90 bar for the investigated material system. Such pressure levels exceed typical autoclave capabilities, indicating that pressure increase alone is not a viable route for achieving structural-grade FVC in current UD staple CF prepreps.

A limitation of the present study is that fibre length was not independently varied. Consequently, while fibre orientation effects on compactibility could be evaluated, the individual influence of fibre length and fibre-length heterogeneity on achievable FVC could not be isolated and therefore should not be interpreted quantitatively from the present dataset. Future work should therefore focus on optimising the carding and alignment processes, as well as exploring strategies to enhance staple CF alignment during part shaping, with the aim of reducing regional FOD variations and identifying an optimal balance between fibre length, alignment quality, and packing efficiency.

Author Contributions: M.P., methodology, formal analysis, writing—original draft preparation, and writing—review and editing; J.K., conceptualization, methodology, formal analysis, validation, writing—original draft preparation, and writing—review and editing; M.H., conceptualization, methodology, formal analysis, writing—review and editing; C.B., supervision, conceptualization, methodology, formal analysis, writing—review and editing; S.B., supervision, writing—review and editing; G.T.G., writing—review and editing. All authors have read and agreed to the published version of the manuscript.

Funding: This research received no external funding.

Data Availability Statement: The original contributions presented in this study are included in the article. Further inquiries can be directed to the corresponding authors.

Acknowledgments: V-Carbon GmbH, Leo Walker, Lucian Zweifel, Véronique Michaud.

Conflicts of Interest: The authors declare no conflicts of interest.

References

1. Such, M.; Ward, C.; Potter, K.D. Aligned Discontinuous Fibre Composites: A Short History. *J. Multifunct. Compos.* **2014**, *2*, 155–168. [[CrossRef](#)]
2. Baz, S.; Ausheyks, L.; Reichert, O.; Dinkelmann, A.; Finckh, H.; Hehl, J.; Poepfel, A.; Gresser, G.T. Recycling of long carbon fibers, Part I: Development of a high aligned RCF-sliver for a binder tape manufacturing process. In Proceedings of the ECCM 2018—18th European Conference on Composite Materials, Athens, Greece, 24–28 June 2018.
3. Goergen, C.; Baz, S.; Mitschang, P.; Gresser, G.T.; Heitmann, U. Plastically deformable thanks to staple fibers. *Kunststoffe Int.* **2016**, *5*, 25–28.
4. Hasan, M.M.B.; Hengstermann, M.; Dilo, R.; Abdkader, A.; Cherif, C. Investigations on the manufacturing and mechanical properties of spun yarns made from staple CF for thermoset composites. *Autex Res. J.* **2017**, *17*, 395–404. [[CrossRef](#)]
5. Xiao, B.; Zaima, T.; Shindo, K.; Kohira, T.; Morisawa, J.; Wan, Y.; Yin, G.; Ohsawa, I.; Takahashi, J. Characterization and elastic property modeling of discontinuous carbon fiber reinforced thermoplastics prepared by a carding and stretching system using treated carbon fibers. *Compos.—A Appl. Sci. Manuf.* **2019**, *126*, 105598. [[CrossRef](#)]
6. Manis, F.; Stegshuster, G.; Wölling, J.; Schlichter, S. Influences on textile and mechanical properties of recycled carbon fiber nonwovens produced by carding. *J. Compos. Sci.* **2021**, *5*, 209. [[CrossRef](#)]
7. Khurshid, M.F.; Abdkader, A.; Cherif, C. Processing of waste carbon and polyamide fibres for high-performance thermoplastic composites: Influence of carding parameters on fibre orientation, fibre length and sliver cohesion force. *J. Text. Inst.* **2020**, *111*, 1277–1287. [[CrossRef](#)]
8. Krauklis, A.E.; Karl, C.W.; Gagani, A.I.; Jørgensen, J.K. Composite material recycling technology—State-of-the-art and sustainable development for the 2020s. *J. Compos. Sci.* **2021**, *5*, 28. [[CrossRef](#)]
9. Zhang, J.; Chevali, V.S.; Wang, H.; Wang, C.H. Current status of carbon fibre and carbon fibre composites recycling. *Compos. B Eng.* **2020**, *193*, 108053. [[CrossRef](#)]
10. Mehdipour, M.; Dogan, S.; Tabrizi, A.T.; Bafqi, M.S.S.; Beylergil, B.; Yildiz, M.; Okan, B.S. Engineering interfacial thermal transport through comparative analysis of electrospraying and dip coating of silanized h-BN for thermo-mechanical enhancement of CF/Epoxy composites. *Compos.—A Appl. Sci. Manuf.* **2025**, *199*, 109264. [[CrossRef](#)]
11. Tutunchi, A.; Ghodrati, T.; Taghizadeh Tabrizi, A.; Osouli-Bostanabad, K. Enhancing the Mechanical Properties of CF-Reinforced Epoxy Composites through Chemically Surface Modification of Carbon Fibers via Novel Two-Step Approach by Addition of Epichlorohydrin. *Funct. Compos. Struct.* **2024**, *6*, 035005. [[CrossRef](#)]

12. Laurencin, T.; Dumont, P.J.J.; Orgéas, L.; Corre, S.L.; Martoia, F.; Rolland du Roscoat, S.; Laure, P. 3D real time and in situ observation of the fibre orientation during the plane strain flow of concentrated fibre suspensions. *J. Nonnewton. Fluid Mech.* **2023**, *312*, 104978. [[CrossRef](#)]
13. Belliveau, R.; Léger, É.; Landry, B.; LaPlante, G. Measuring fibre orientation and predicting elastic properties of discontinuous long fibre thermoplastic composites. *J. Compos. Mater.* **2021**, *55*, 321–330. [[CrossRef](#)]
14. Sebaey, T.A.; Catalanotti, G.; O'Dowd, N.P. A microscale integrated approach to measure and model fibre misalignment in fibre-reinforced composites. *Compos. Sci. Technol.* **2019**, *183*, 107793. [[CrossRef](#)]
15. Zweifel, L.; Kupski, J.; Dransfeld, C.; Caglar, B.; Baz, S.; Cessario, D.; Gresser, G.T.; Brauner, C. Multiscale Characterisation of Staple Carbon Fibre-Reinforced Polymers. *J. Compos. Sci.* **2023**, *7*, 465. [[CrossRef](#)]
16. Salaberger, D.; Kannappan, K.A.; Kastner, J.; Reussner, J.; Auinger, T. Evaluation of computed tomography data from fibre reinforced polymers to determine fibre length distribution. *Int. Polym. Process.* **2011**, *26*, 283–291. [[CrossRef](#)]
17. Melenka, G.W.; Gholami, A. Fiber identification of braided composites using micro-computed tomography. *Compos. Commun.* **2021**, *27*, 100813. [[CrossRef](#)]
18. Maksimcuka, J.; Obata, A.; Sampson, W.W.; Blanc, R.; Gao, C.; Withers, P.J.; Tsigkou, O.; Kasuga, T.; Lee, P.D.; Poologasundarampillai, G. X-ray tomographic imaging of tensile deformation modes of electrospun biodegradable polyester fibers. *Front. Mater.* **2017**, *4*, 43. [[CrossRef](#)]
19. Diaz, A.; Guizar-Sicairos, M.; Poeppel, A.; Menzel, A.; Bunk, O. Characterization of carbon fibers using X-ray phase nanotomography. *Carbon* **2014**, *67*, 98–103. [[CrossRef](#)]
20. Gomasasca, S.; Peeters, D.M.J.; Atli-Veltin, B.; Dransfeld, C. Characterising microstructural organisation in unidirectional composites. *Compos. Sci. Technol.* **2021**, *215*, 109030. [[CrossRef](#)]
21. Katuin, N.; Peeters, D.M.J.; Dransfeld, C.A. Method for the microstructural characterisation of unidirectional composite tapes. *J. Compos. Sci.* **2021**, *5*, 275. [[CrossRef](#)]
22. Nikishkov, Y.; Seon, G.; Makeev, A. Structural analysis of composites with porosity defects based on X-ray computed tomography. *J. Compos. Mater.* **2014**, *48*, 2131–2144. [[CrossRef](#)]
23. Nikishkov, Y.; Airoidi, L.; Makeev, A. Measurement of voids in composites by X-ray Computed Tomography. *Compos. Sci. Technol.* **2013**, *89*, 89–97. [[CrossRef](#)]
24. Yu, H.; Potter, K.D.; Wisnom, M.R. A novel manufacturing method for aligned discontinuous fibre composites (High Performance-Discontinuous Fibre method). *Compos.–A Appl. Sci. Manuf.* **2014**, *65*, 175–185. [[CrossRef](#)]
25. Sharma, B.N.; Naragani, D.; Nguyen, B.N.; Tucker, C.L.; Sangid, M.D. Uncertainty quantification of fiber orientation distribution measurements for long-fiber-reinforced thermoplastic composites. *J. Compos. Mater.* **2018**, *52*, 1781–1797. [[CrossRef](#)]
26. Longana, M.L.; Ong, N.; Yu, H.N.; Potter, K.D. Multiple closed loop recycling of carbon fibre composites with the HiPerDiF (High Performance Discontinuous Fibre) method. *Compos. Struct.* **2016**, *153*, 271–277. [[CrossRef](#)]
27. Yarlagadda, S.; Deitzel, J.; Heider, D.; Tierney, J.; Gillespie, J.W. Tailorable Universal Feedstock for Forming (TUFF): Overview and performance. In Proceedings of the International SAMPE Technical Conference, Charlotte, NC, USA, 20–23 May 2019.
28. Yamanaka, A.; Terada, M.; Ichiki, M.; Kimoto, Y.; Shiraki, K.; Nagata, M.; Shimamoto, D.; Hotta, Y. Evaluation of fiber orientation by x-ray diffraction on carbon fiber reinforced polyamide 6. *J. Fiber Sci. Technol.* **2020**, *76*, 199–207. [[CrossRef](#)]
29. Yang, X.; Ju, B.F.; Kersemans, M. Assessment of the 3D ply-by-ply fiber structure in impacted CFRP by means of planar Ultrasound Computed Tomography (pU-CT). *Compos. Struct.* **2022**, *279*, 114745. [[CrossRef](#)]
30. Pelivanov, I.; Ambroziński, Ł.; Khomenko, A.; Koricho, E.G.; Cloud, G.L.; Haq, M.; O'Donnell, M. High resolution imaging of impacted CFRP composites with a fiber-optic laser-ultrasound scanner. *Photoacoustics* **2016**, *4*, 55–64. [[CrossRef](#)]
31. Dong, L.; Hang, H.; Park, J.G.; Mio, W.; Liang, R. Detecting Carbon Nanotube Orientation with Topological Analysis of Scanning Electron Micrographs. *Nanomaterials* **2022**, *12*, 1251. [[CrossRef](#)]
32. Alavi, S.H.; Ruiz, V.; Krasieva, T.; Botvinick, E.L.; Kheradvar, A. Characterizing the collagen fiber orientation in pericardial leaflets under mechanical loading conditions. *Ann. Biomed. Eng.* **2013**, *41*, 547–561. [[CrossRef](#)] [[PubMed](#)]
33. Huntley, S.J.; Rendall, T.; Longana, M.L.; Pozegic, T.; Potter, K.D.; Hamerton, I. Validation of a Smoothed Particle Hydrodynamics Model for a Highly Aligned Discontinuous Fibre Composites Manufacturing Process. *Compos. Sci. Technol.* **2020**, *196*, 108152. [[CrossRef](#)]
34. Kirupanantham, G. Characterisation of Discontinuous Carbon Fibre Preforms for Automotive Applications. Ph.D. Thesis, University of Nottingham, Nottingham, UK, 2013.
35. Cox, H.L. The elasticity and strength of paper and other fibrous materials. *Br. J. Appl. Phys.* **1952**, *3*, 72–79. [[CrossRef](#)]
36. Edwards, H.; Evans, N.P. A method for the production of high quality aligned short fibre mats and their composites. In Proceedings of the Advances in Composite Materials, 3rd International Conference on Composite Materials, Paris, France, 26–29 August 1980; pp. 1620–1635.
37. Eom, Y.; Boogh, L.; Michaud, V.; Manson, J.-A. A Structure and Property Based Process Window for Void Free Thermoset Composites. *Polym. Compos.* **2001**, *22*, 22–31. [[CrossRef](#)]

38. Hubert, P.; Poursartip, A. Method for the direct measurement of the fibre bed compaction curve of composite prepregs. *Compos.–A Appl. Sci. Manuf.* **2001**, *32*, 179–187. [[CrossRef](#)]
39. Servais, C.; Michaud, V.; Mansons, J.-A. The Packing Stress of Impregnated Fiber Mats. *Polym. Compos.* **2001**, *22*, 298–311. [[CrossRef](#)]
40. Merhi, D.; Comte, E.; Michaud, V.; Mansons, J.A. Correlation Between Sizing Formulation and Compressive Behavior of Concentrated Glass Bundle Suspensions. *Polym. Compos.* **2005**, *26*, 370–376. [[CrossRef](#)]
41. May, D.; Kühn, F.; Etchells, M.; Fauster, E.; Endruweit, A.; Lira, C. A reference specimen for compaction tests of fiber reinforcements. *Adv. Manuf. Polym. Compos. Sci.* **2019**, *5*, 230–233. [[CrossRef](#)]
42. Sousa, P.; Lomov, S.; Ivens, J. Methodology of dry and wet compressibility measurement. *Compos.–A Appl. Sci. Manuf.* **2020**, *128*, 105672. [[CrossRef](#)]
43. Bender, M.; Fauser, E. Novel test-rig for compaction behaviour analysis of textile reinforcements for improved RTM-process replication. *Adv. Manuf. Polym. Compos. Sci.* **2023**, *9*, 2263828. [[CrossRef](#)]
44. Grieder, S.; Zhilyaev, I.; Küng, M.; Brauner, C.; Akermann, M.; Bosshard, J.; Inderkum, P.; Francisco, J.; Willemin, Y.; Eichenhofer, M. Consolidation of Additive Manufactured Continuous Carbon Fiber Reinforced Polyamide 12 Composites and the Development of Process-Related Numerical Simulation Methods. *Polymers* **2022**, *14*, 3429. [[CrossRef](#)]
45. Walker, L. Development of Prepreg and Processing Technologies for recycled Carbon Fibre Based Textiles. Master's Thesis, FHNW University of Applied Sciences and Arts Northwestern Switzerland, Windisch, Switzerland, 2021.

Disclaimer/Publisher's Note: The statements, opinions and data contained in all publications are solely those of the individual author(s) and contributor(s) and not of MDPI and/or the editor(s). MDPI and/or the editor(s) disclaim responsibility for any injury to people or property resulting from any ideas, methods, instructions or products referred to in the content.